

## **PUMA-560 Robot Manipulator Position Computed Torque Control Methods Using MATLAB/SIMULINK and Their Integration into Graduate Nonlinear Control and MATLAB Courses**

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### **Abstract**

This paper describes the MATLAB/SIMULINK realization of the PUMA 560 robot manipulator position control methodology. This paper focuses on design, analyzed and implements nonlinear computed torque control (CTC) methods. These simulation models are developed as a part of a software laboratory to support and enhance graduate/undergraduate robotics courses, nonlinear control courses and MATLAB/SIMULINK courses at research and development company (SSP Co.) research center, Shiraz, Iran.

**Keywords:** MATLAB/SIMULINK, PUMA 560 Robot Manipulator, Position Control Method, Computed Torque Control, Robotics, Nonlinear Control.

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## 1. INTRODUCTION

Computer modeling, simulation and implementation tools have been widely used to support and develop nonlinear control, robotics, and MATLAB/SIMULINK courses. MATLAB with its toolboxes such as SIMULINK [1] is one of the most accepted software packages used by researchers to enhance teaching the transient and steady-state characteristics of control and robotic courses [3\_7]. In an effort to modeling and implement robotics, nonlinear control and advanced MATLAB/SIMULINK courses at research and development SSP Co., authors have developed MATLAB/SIMULINK models for learn the basic information in field of nonlinear control and industrial robot manipulator [8, 9].

Controller is a device which can sense information from linear or nonlinear system (e.g., robot manipulator) to improve the systems performance [3]. The main targets in designing control systems are stability, good disturbance rejection, and small tracking error[5]. Several industrial robot manipulators are controlled by linear methodologies (e.g., Proportional-Derivative (PD) controller, Proportional- Integral (PI) controller or Proportional- Integral-Derivative (PID) controller), but when robot manipulator works with various payloads and have uncertainty in dynamic models this technique has limitations. From the control point of view, uncertainty is divided into two main groups: uncertainty in unstructured inputs (e.g., noise, disturbance) and uncertainty in structure dynamics (e.g., payload, parameter variations). In some applications robot manipulators are used in an unknown and unstructured environment, therefore strong mathematical tools used in new control methodologies to design nonlinear robust controller with an acceptable performance (e.g., minimum error, good trajectory, disturbance rejection). Joint space and operational space control are closed loop controllers which they have been used to provide robustness and rejection of disturbance effect. The main target in joint space controller is design a feedback controller that allows the actual motion ( $q_a(t)$ ) tracking of the desired motion ( $q_d(t)$ ). This control problem is classified into two main groups. Firstly, transformation the desired motion  $X_d(t)$  to joint variable  $q_d(t)$  by inverse kinematics of robot manipulators [6]. Figure 1 shows the main block diagram of joint space controller. The main target in operational space controller is to design a feedback controller to allow the actual end-effector motion  $X_a(t)$  to track the desired endeffector motion  $X_d(t)$ . This control methodology requires a greater algorithmic complexity and the inverse kinematics used in the feedback control loop. Direct measurement of operational space variables are very expensive that caused to limitation used of this controller in industrial robot manipulators[6]. Figure 2 shows the main block diagram of operational space control.

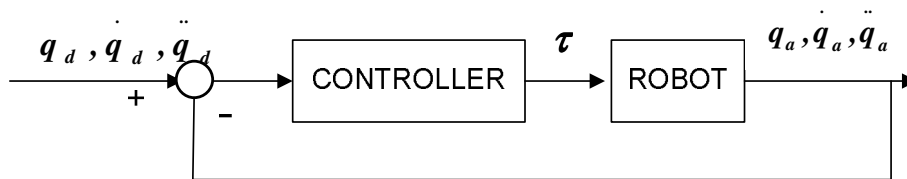


FIGURE 1: Block diagram of joint space control

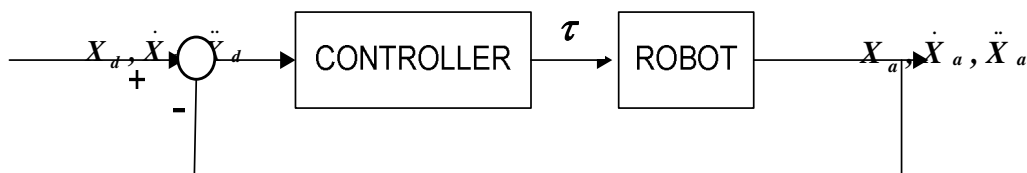


FIGURE 2: Block diagram of operational space control

Computed torque controller (CTC) is a powerful nonlinear controller which it widely used in control of robot manipulator. It is based on feedback linearization and computes the required arm torques using the nonlinear feedback control law. This controller works very well when all dynamic and physical parameters are known but when the robot manipulator has variation in dynamic parameters, in this situation the controller has no acceptable performance[14]. In practice, most of physical systems (e.g., robot manipulators) parameters are unknown or time variant, therefore, computed torque like controller used to compensate dynamic equation of robot manipulator[1, 6]. Research on computed torque controller is significantly growing on robot manipulator application which has been reported in [1, 6, 15-16]. Vivas and Mosquera [15] have proposed a predictive functional controller and compare to computed torque controller for tracking response in uncertain environment. However both controllers have been used in feedback linearization, but predictive strategy gives better result as a performance. A computed torque control with non parametric regression models have been presented for a robot arm[16]. This controller also has been problem in uncertain dynamic models. Based on [1, 6] and [15-16] computed torque controller is a significant nonlinear controller to certain systems which it is based on feedback linearization and computes the required arm torques using the nonlinear feedback control law.

This paper is organized as follows:

In section 2, dynamic formulation of robot manipulator is presented. Detail of classical CTC and MATLAB/SIMULINK implementation of this controller is presented in section 3. In section 4, the simulation result is presented and finally in section 5, the conclusion is presented.

## 2. PUMA 560 ROBOT MANIPULATOR DYNAMIC FORMULATION

**Dynamics of PUMA560 Robot Manipulator:** To position control of robot manipulator, the second three axes are locked the dynamic equation of PUMA robot manipulator is given by [77-80];

$$M(\theta) \begin{bmatrix} \ddot{\theta}_1 \\ \ddot{\theta}_2 \\ \ddot{\theta}_3 \end{bmatrix} + B(\theta) \begin{bmatrix} \dot{\theta}_1 \dot{\theta}_2 \\ \dot{\theta}_1 \dot{\theta}_3 \\ \dot{\theta}_2 \dot{\theta}_3 \end{bmatrix} + C(\theta) \begin{bmatrix} \dot{\theta}_1^2 \\ \dot{\theta}_2^2 \\ \dot{\theta}_3^2 \end{bmatrix} + G(\theta) = \begin{bmatrix} \tau_1 \\ \tau_2 \\ \tau_3 \end{bmatrix} \quad (1)$$

Where

$$M(q) = \begin{bmatrix} M_{11} & M_{12} & M_{13} & 0 & 0 & 0 \\ M_{21} & M_{22} & M_{23} & 0 & 0 & 0 \\ M_{31} & M_{32} & M_{33} & 0 & M_{35} & 0 \\ 0 & 0 & 0 & M_{44} & 0 & 0 \\ 0 & 0 & 0 & 0 & M_{55} & 0 \\ 0 & 0 & 0 & 0 & 0 & M_{66} \end{bmatrix} \quad (2)$$

$M$  is computed as

$$M_{11} = I_{m1} + I_1 + I_3 \times \cos(\theta_2) \cos(\theta_2) + I_7 \sin(\theta_2 + \theta_3) \sin(\theta_2 + \theta_3) + I_{10} \sin(\theta_2 + \theta_3) \cdot I_{11} \sin(\theta_2) \cos(\theta_2) + I_{21} \sin(\theta_2 + \theta_3) \sin(\theta_2 + \theta_3) + 2 + [I_5 \cos(\theta_2) \sin(\theta_2 + \theta_3) + I_{12} \cos(\theta_2) \cos(\theta_2 + \theta_3) + I_{15} \sin(\theta_2 + \theta_3) \sin(\theta_2 + \theta_3) + I_{16} \cos(\theta_2) \sin(\theta_2 + \theta_3) \theta_3] \cos(\theta_2 + \theta_3) \quad (3)$$

$$M_{12} = I_4 \sin(\theta_2) + I_8 \cos(\theta_2 + \theta_3) + I_9 \cos(\theta_2) + I_{13} \sin(\theta_2 + \theta_3) - I_{18} \cos(\theta_2 + \theta_3) \quad (4)$$

$$M_{13} = I_8 \cos(\theta_2 + \theta_3) + I_{13} \sin(\theta_2 + \theta_3) - I_{18} \cos(\theta_2 + \theta_3) \quad (5)$$

$$M_{22} = I_{m2} + I_2 + I_6 + 2[I_5 \sin(\theta_3) + I_{12} \cos(\theta_2)] + I_{15} + I_{16} \sin(\theta_3) \quad (6)$$

$$M_{23} = I_5 \sin(\theta_3) + I_6 + I_{12} \cos(\theta_3) + I_{16} \sin(\theta_3) + 2I_{15} \quad (7)$$

$$M_{33} = I_{m3} + I_6 + 2I_{15} \quad (8)$$

$$M_{35} = I_{15} + I_{17} \quad (9)$$

$$M_{44} = I_{m4} + I_{14} \quad (10)$$

$$M_{55} = I_{m5} + I_{17} \quad (11)$$

$$M_{66} = I_{m6} + I_{23} \quad (12)$$

$$M_{21} = M_{12} , M_{31} = M_{13} \text{ and } M_{32} = M_{23} \quad (13)$$

and Coriolis ( $B$ ) matrix is calculated as the following

$$B(q) = \begin{bmatrix} b_{112} & b_{113} & 0 & b_{115} & 0 & b_{123} & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & b_{214} & 0 & 0 & b_{223} & 0 & b_{225} & 0 & 0 & b_{235} & 0 & 0 & 0 \\ 0 & 0 & b_{314} & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ b_{412} & b_{412} & 0 & b_{415} & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & b_{514} & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \end{bmatrix} \quad (14)$$

Where,

$$b_{112} = 2[-I_3 \sin(\theta_2) \cos(\theta_2) + I_5 \cos(\theta_2 + \theta_2 + \theta_3) + I_7 \sin(\theta_2 + \theta_3) \cos(\theta_2 + \theta_3) \theta_2 + \theta_3) - I_{15} 2 \sin(\theta_2 + \theta_3) \cos(\theta_2 + \theta_3) + I_{16} \cos(\theta_2 + \theta_2 + \theta_3) + I_{21} \sin(\theta_2 + \theta_3) + I_{22} (1 - 2 \sin(\theta_2 + \theta_3) \sin(\theta_2 + \theta_3))] + I_{10} (1 - 2 \sin(\theta_2 + \theta_3) \sin(\theta_2 + \theta_3)) 2 \sin(\theta_2) \sin(\theta_2)] \quad (15)$$

$$b_{113} = 2[I_5 \cos(\theta_2) \cos(\theta_2 + \theta_3) + I_7 \sin(\theta_2 + \theta_3) \cos(\theta_2 + \theta_3) - I_{12} \cos(\theta_2) \sin(\theta_2 + \theta_3) + I_{15} 2 \sin(\theta_2 + \theta_3) \cos(\theta_2 + \theta_3) + I_{16} \cos(\theta_2) \cos(\theta_2 + \theta_3) + I_{21} \sin(\theta_2 + \theta_3) \cos(\theta_2 + \theta_3) + I_{22} (1 - 2 \sin(\theta_2 + \theta_3) \sin(\theta_2 + \theta_3))] + I_{10} (1 - 2 \sin(\theta_2 + \theta_3) \sin(\theta_2 + \theta_3)) \quad (16)$$

$$b_{115} = 2[-\sin(\theta_2 + \theta_3) \cos(\theta_2 + \theta_3) + I_{15} 2 \sin(\theta_2 + \theta_3) \cos(\theta_2 + \theta_3) + I_{16} \cos(\theta_2 + \theta_3) + I_{22} \cos(\theta_2 + \theta_3) \cos(\theta_2 + \theta_3)] \quad (17)$$

$$b_{123} = 2[-I_8 \sin(\theta_2 + \theta_3) + I_{13} \cos(\theta_2 + \theta_3) + I_{18} \sin(\theta_2 + \theta_3)] \quad (18)$$

$$b_{214} = I_{14} \sin(\theta_2 + \theta_3) + I_{19} \sin(\theta_2 + \theta_3) + 2I_{20} \sin(\theta_2 + \theta_3) (1 - 0.5) \quad (19)$$

$$b_{223} = 2[-I_{12}\sin(\theta_3) + I_5\cos(\theta_3) + I_{16}\cos(\theta_3)] \quad (20)$$

$$b_{235} = 2[I_{16}\cos(\theta_3) + I_{22}] \quad (21)$$

$$b_{314} = 2[I_{20}\sin(\theta_2 + \theta_3)(1 - 0.5)] + I_{14}\sin(\theta_2 + \theta_3) + I_{19}\sin(\theta_2 + \theta_3) \quad (22)$$

$$b_{412} = b_{214} = -[I_{14}\sin(\theta_2 + \theta_3) + I_{19}\sin(\theta_2 + \theta_3) + 2I_{20}\sin(\theta_2 + \theta_3)(1 - 0.5)] \quad (23)$$

$$b_{413} = -b_{314} = -2[I_{20}\sin(\theta_2 + \theta_3)(1 - 0.5)] + I_{14}\sin(\theta_2 + \theta_3) + I_{19}\sin(\theta_2 + \theta_3) \quad (24)$$

$$b_{415} = -I_{20}\sin(\theta_2 + \theta_3) - I_{17}\sin(\theta_2 + \theta_3) \quad (25)$$

$$b_{514} = -b_{415} = I_{20}\sin(\theta_2 + \theta_3) + I_{17}\sin(\theta_2 + \theta_3) \quad (26)$$

consequently coriolis matrix is shown as bellows;

$$B(q) \cdot \ddot{q} = \begin{bmatrix} b_{112} \cdot q_1 \ddot{q}_2 + b_{113} \cdot q_1 \ddot{q}_3 + 0 + b_{123} \cdot \ddot{q}_2 \ddot{q}_3 \\ 0 + b_{223} \cdot \ddot{q}_2 \ddot{q}_3 + 0 + 0 \\ 0 \\ b_{412} \cdot q_1 \ddot{q}_2 + b_{413} \cdot q_1 \ddot{q}_3 + 0 \\ 0 \\ 0 \end{bmatrix} \quad (27)$$

Moreover Centrifugal (C) matrix is demonstrated as

$$C(q) = \begin{bmatrix} 0 & C_{12} & C_{13} & 0 & 0 & 0 \\ C_{21} & 0 & C_{23} & 0 & 0 & 0 \\ C_{31} & C_{32} & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ C_{51} & C_{52} & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \end{bmatrix} \quad (28)$$

Where,

$$c_{12} = I_4\cos(\theta_2) - I_8\sin(\theta_2 + \theta_3) - I_9\sin(\theta_2) + I_{13}\cos(\theta_2 + \theta_3) + I_{18}\sin(\theta_2 + \theta_3) \quad (29)$$

$$c_{13} = 0.5b_{123} = -I_8\sin(\theta_2 + \theta_3) + I_{13}\cos(\theta_2 + \theta_3) + I_{18}\sin(\theta_2 + \theta_3) \quad (30)$$

$$c_{21} = -0.5b_{112} = I_3\sin(\theta_2)\cos(\theta_2) - I_5\cos(\theta_2 + \theta_2 + \theta_3) - I_7\sin(\theta_2 + \theta_3)\cos(\theta_2 + \theta_2 + \theta_3) + I_{15}2\sin(\theta_2 + \theta_3)\cos(\theta_2 + \theta_3) - I_{16}\cos(\theta_2 + \theta_2 + \theta_3) - I_{21}\sin(\theta_2 + \theta_3)\cos(\theta_2 + \theta_3) - I_{22}(1 - 2\sin(\theta_2 + \theta_3)\sin(\theta_2 + \theta_3)) - 0.5I_{10}(1 - 2\sin(\theta_2 + \theta_3)\sin(\theta_2)) - 0.5I_{11}(1 - 2\sin(\theta_2)\sin(\theta_2)) \quad (31)$$

$$c_{22} = 0.5b_{223} = -I_{12}\sin(\theta_3) + I_5\cos(\theta_3) + I_{16}\cos(\theta_3) \quad (32)$$

$$c_{23} = -0.5b_{113} = -I_5\cos(\theta_2)\cos(\theta_2 + \theta_3) - I_7\sin(\theta_2 + \theta_3)\cos(\theta_2 + \theta_3) + I_{12}\sin(\theta_2) - I_{15}2\sin(\theta_2 + \theta_3)\cos(\theta_2 + \theta_3) - I_{16}\cos(\theta_2)\cos(\theta_2 + \theta_3) - I_{21}\sin(\theta_2 + \theta_3)\cos(\theta_2 + \theta_3) - I_{22}(1 - 2\sin(\theta_2 + \theta_3)\sin(\theta_2 + \theta_3)) - 0.5I_{10}(1 - 2\sin(\theta_2 + \theta_3)\sin(\theta_2 + \theta_3)) \quad (33)$$

$$c_{31} = -c_{23} = I_{12} \sin(\theta_3) - I_5 \cos(\theta_3) - I_{16} \cos(\theta_3) \quad (34)$$

$$c_{32} = -0.5 b_{115} = \sin(\theta_2 + \theta_3) \cos(\theta_2 + \theta_3) - I_{15} 2 \sin(\theta_2 + \theta_3) \cos(\theta_2 + \theta_3) - I_{16} \cos(\theta_2) \cos(\theta_2 + \theta_3) + I_{22} \cos(\theta_2 + \theta_3) \cos(\theta_2 + \theta_3) \quad (35)$$

$$c_{52} = -0.5 b_{225} = -I_{16} \cos(\theta_3) - I_{22} \quad (36)$$

In this research  $q_4 = q_5 = q_6 = 0$  , as a result

$$C(q) \cdot \dot{q}^2 = \begin{bmatrix} c_{112} \cdot q_2^2 + c_{13} \cdot q_3^2 \\ c_{21} \cdot q_1^2 + c_{23} \cdot q_3^2 \\ c_{13} \cdot q_1^2 + c_{32} \cdot q_2^2 \\ 0 \\ c_{51} \cdot q_1^2 + c_{52} \cdot q_2^2 \\ 0 \end{bmatrix} \quad (37)$$

Gravity ( $G$ ) Matrix can be written as

$$G(q) = \begin{bmatrix} 0 \\ g_2 \\ g_3 \\ 0 \\ g_5 \\ 0 \end{bmatrix} \quad (38)$$

Where,

$$G_2 = g_1 \cos(\theta_2) + g_2 \sin(\theta_2 + \theta_3) + g_3 \sin(\theta_2) + g_4 \cos(\theta_2 + \theta_3) + g_5 \sin(\theta_2 + \theta_3) \quad (39)$$

$$G_3 = g_2 \sin(\theta_2 + \theta_3) + g_4 \cos(\theta_2 + \theta_3) + g_5 \sin(\theta_2 + \theta_3) \quad (40)$$

$$G_5 = g_5 \sin(\theta_2 + \theta_3) \quad (41)$$

Suppose  $\ddot{q}$  is written as follows

$$\ddot{q} = M^{-1}(q) \cdot \{\tau - [B'(q) \dot{q} \dot{q} + C(q) \dot{q}^2 + g(q)]\} \quad (42)$$

and  $K$  is introduced as

$$K = \{\tau - [B'(q) \dot{q} \dot{q} + C(q) \dot{q}^2 + g(q)]\} \quad (43)$$

$\ddot{q}$  can be written as

$$\ddot{q} = M^{-1}(q) \cdot K \quad (44)$$

Therefore  $K$  for PUMA robot manipulator is calculated by the following equations

$$K_1 = \tau_1 - [b_{112} \dot{q}_1 \dot{q}_2 + b_{113} \dot{q}_1 \dot{q}_3 + 0 + b_{123} \dot{q}_2 \dot{q}_3] - [C_{12} \dot{q}_2^2 + C_{13} \dot{q}_3^2] - g_1 \quad (45)$$

$$K_2 = \tau_2 - [b_{223} \dot{q}_2 \dot{q}_3] - [C_{21} \dot{q}_1^2 + C_{23} \dot{q}_3^2] - g_2 \quad (46)$$

$$K_3 = \tau_3 - [C_{31}\dot{q}_1^2 + C_{32}\dot{q}_2^2] - g_3 \tag{47}$$

$$K_4 = \tau_4 - [b_{412}\dot{q}_1\dot{q}_2 + b_{413}\dot{q}_1\dot{q}_3] - g_4 \tag{48}$$

$$K_5 = \tau_5 - [C_{51}\dot{q}_1^2 + C_{52}\dot{q}_2^2] - g_5 \tag{49}$$

$$K_6 = \tau_6 \tag{50}$$

An information about inertial constant and gravitational constant are shown in Tables 1 and 2 based on the studies carried out by Armstrong [80] and Corke and Armstrong [81].

$I_1 = 1.43 \pm 0.05$	$I_2 = 1.75 \pm 0.07$
$I_3 = 1.38 \pm 0.05$	$I_4 = 0.69 \pm 0.02$
$I_5 = 0.372 \pm 0.031$	$I_6 = 0.333 \pm 0.016$
$I_7 = 0.298 \pm 0.029$	$I_8 = -0.134 \pm 0.014$
$I_9 = 0.0238 \pm 0.012$	$I_{10} = -0.0213 \pm 0.0022$
$I_{11} = -0.0142 \pm 0.0070$	$I_{12} = -0.011 \pm 0.0011$
$I_{13} = -0.00379 \pm 0.0009$	$I_{14} = 0.00164 \pm 0.000070$
$I_{15} = 0.00125 \pm 0.0003$	$I_{16} = 0.00124 \pm 0.0003$
$I_{17} = 0.000642 \pm 0.0003$	$I_{18} = 0.000431 \pm 0.00013$
$I_{19} = 0.0003 \pm 0.0014$	$I_{20} = -0.000202 \pm 0.0008$
$I_{21} = -0.0001 \pm 0.0006$	$I_{22} = -0.000058 \pm 0.00001$
$I_{23} = 0.00004 \pm 0.00002$	$I_{m1} = 1.14 \pm 0.27$
$I_{m2} = 4.71 \pm 0.54$	$I_{m3} = 0.827 \pm 0.093$
$I_{m4} = 0.2 \pm 0.016$	$I_{m5} = 0.179 \pm 0.014$
$I_{m6} = 0.193 \pm 0.016$	

TABLE 1: Inertial constant reference (Kg.m<sup>2</sup>)

$g_1 = -37.2 \pm 0.5$	$g_2 = -8.44 \pm 0.20$
$g_3 = 1.02 \pm 0.50$	$g_4 = 0.249 \pm 0.025$
$g_5 = -0.0282 \pm 0.0056$	

TABLE 2: Gravitational constant (N.m)

### 3. CONTROL: COMPUTED TORQUE CONTROLLER ANALYSIS, MODELLING AND IMPLEMENTATION ON PUMA 560 ROBOT MANIPULATOR

Computed torque controller (CTC) is a powerful nonlinear controller which it widely used in control of robot manipulator. It is based on feedback linearization and computes the required arm torques using the nonlinear feedback control law. This controller works very well when all dynamic and physical parameters are known but when the robot manipulator has variation in dynamic parameters, in this situation the controller has no acceptable performance[14]. In practice, most of physical systems (e.g., robot manipulators) parameters are unknown or time variant, therefore, computed torque like controller used to compensate dynamic equation of robot manipulator[1, 6]. Research on computed torque controller is significantly growing on robot manipulator application which has been reported in [1, 6, 15-16]. Vivas and Mosquera [15] have proposed a predictive functional controller and compare to computed torque controller for tracking response in uncertain environment. However both controllers have been used in feedback linearization, but predictive strategy gives better result as a performance. A computed torque control with non parametric regression models have been presented for a robot arm[16]. This controller also has been problem in uncertain dynamic models. Based on [1, 6] and [15-16] computed torque controller is a significant nonlinear controller to certain systems which it is based on feedback linearization and computes the required arm torques using the nonlinear feedback control law. When all dynamic and physical parameters are known, computed torque controller works fantastically; practically a large amount of systems have uncertainties, therefore sliding mode controller is one of the best case to solve this challenge.

The central idea of Computed torque controller (CTC) is feedback linearization so, originally this algorithm is called feedback linearization controller. It has assumed that the desired motion trajectory for the manipulator  $q_d(t)$ , as determined, by a path planner. Defines the tracking error as:

$$e(t) = q_d(t) - q_a(t) \quad (51)$$

Where  $e(t)$  is error of the plant,  $q_d(t)$  is desired input variable, that in our system is desired displacement,  $q_a(t)$  is actual displacement. If an alternative linear state-space equation in the form  $\dot{x} = Ax + BU$  can be defined as

$$\dot{x} = \begin{bmatrix} 0 & I \\ 0 & 0 \end{bmatrix} x + \begin{bmatrix} 0 \\ I \end{bmatrix} U \quad (52)$$

With  $U = -M^{-1}(q) \cdot N(q, \dot{q}) + M^{-1}(q) \cdot \tau$  and this is known as the Brunousky canonical form. By equation (51) and (52) the Brunousky canonical form can be written in terms of the state  $x = [e^T \ \dot{e}^T]^T$  as [1]:

$$\frac{d}{dt} \begin{bmatrix} e \\ \dot{e} \end{bmatrix} = \begin{bmatrix} 0 & I \\ 0 & 0 \end{bmatrix} \cdot \begin{bmatrix} e \\ \dot{e} \end{bmatrix} + \begin{bmatrix} 0 \\ I \end{bmatrix} U \quad (53)$$

With

$$U = \ddot{q}_d + M^{-1}(q) \cdot \{N(q, \dot{q}) - \tau\} \quad (54)$$

Then compute the required arm torques using inverse of equation (55), is;

$$\tau = M(q)(\ddot{q}_d - U) + N(q, \dot{q}) \quad (55)$$

This is a nonlinear feedback control law that guarantees tracking of desired trajectory. Selecting proportional-plus-derivative (PD) feedback for  $U(t)$  results in the PD-computed torque controller [6];

$$\tau = M(q)(\ddot{q}_d + K_v \dot{e} + K_p e) + N(q, \dot{q}) \quad (56)$$



and the resulting linear error dynamics are

$$(\ddot{q}_d + K_v \dot{e} + K_p e) = 0 \quad (57)$$

According to the linear system theory, convergence of the tracking error to zero is guaranteed [6]. Where  $K_p$  and  $K_v$  are the controller gains. The result schemes is shown in Figure 3, in which two feedback loops, namely, inner loop and outer loop, which an inner loop is a compensate loop and an outer loop is a tracking error loop.

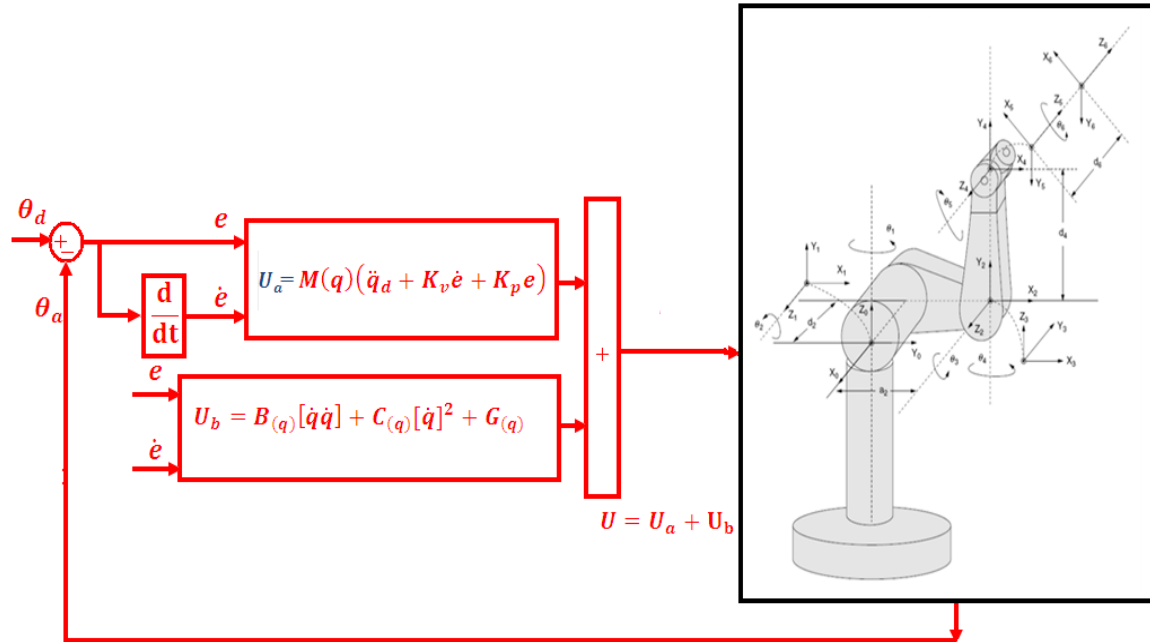


FIGURE 3: Block diagram of PD-computed torque controller (PD-CTC)

The application of proportional-plus-derivative (PD) computed torque controller to control of PUMA 560 robot manipulator introduced in this part. Suppose that in (58) the nonlinearity term defined by the following term

$$N(q, \dot{q}) = B(q)\dot{q}\dot{q} + C(q)\dot{q}^2 + g(q) \quad (58)$$

$$\begin{bmatrix} b_{112}\dot{q}_1\dot{q}_2 + b_{113}\dot{q}_1\dot{q}_3 + 0 + b_{123}\dot{q}_2\dot{q}_3 \\ 0 + b_{223}\dot{q}_2\dot{q}_3 + 0 + 0 \\ 0 \\ b_{412}\dot{q}_1\dot{q}_2 + b_{413}\dot{q}_1\dot{q}_3 + 0 + 0 \\ 0 \\ 0 \end{bmatrix} + \begin{bmatrix} C_{12}\dot{q}_2^2 + C_{13}\dot{q}_3^2 \\ C_{21}\dot{q}_1^2 + C_{23}\dot{q}_3^2 \\ C_{31}\dot{q}_1^2 + C_{32}\dot{q}_2^2 \\ 0 \\ C_{51}\dot{q}_1^2 + C_{52}\dot{q}_2^2 \\ 0 \end{bmatrix} + \begin{bmatrix} 0 \\ g_2 \\ g_3 \\ 0 \\ g_5 \\ 0 \end{bmatrix}$$

Therefore the equation of PD-CTC for control of PUMA 560 robot manipulator is written as the equation of (59);

$$\begin{bmatrix} \ddot{q}_1 \\ \ddot{q}_2 \\ \ddot{q}_3 \\ \ddot{q}_4 \\ \ddot{q}_5 \\ \ddot{q}_6 \end{bmatrix} = \begin{bmatrix} M_{11} & M_{12} & M_{13} & 0 & 0 & 0 \\ M_{21} & M_{22} & M_{23} & 0 & 0 & 0 \\ M_{31} & M_{32} & M_{33} & 0 & M_{35} & 0 \\ 0 & 0 & 0 & M_{44} & 0 & 0 \\ 0 & 0 & 0 & 0 & M_{55} & 0 \\ 0 & 0 & 0 & 0 & 0 & M_{66} \end{bmatrix} \begin{bmatrix} \ddot{q}_{d1} + K_{v1}\dot{e}_1 + K_{p1}e_1 \\ \ddot{q}_{d2} + K_{v2}\dot{e}_2 + K_{p2}e_2 \\ \ddot{q}_{d3} + K_{v3}\dot{e}_3 + K_{p3}e_3 \\ \ddot{q}_{d4} + K_{v4}\dot{e}_4 + K_{p4}e_4 \\ \ddot{q}_{d5} + K_{v5}\dot{e}_5 + K_{p5}e_5 \\ \ddot{q}_{d6} + K_{v6}\dot{e}_6 + K_{p6}e_6 \end{bmatrix} \quad (59)$$

$$+ \begin{bmatrix} b_{112} \dot{q}_1 \dot{q}_2 + b_{113} \dot{q}_1 \dot{q}_3 + 0 + b_{123} \dot{q}_2 \dot{q}_3 \\ 0 + b_{223} \dot{q}_2 \dot{q}_3 + 0 + 0 \\ 0 \\ b_{412} \dot{q}_1 \dot{q}_2 + b_{413} \dot{q}_1 \dot{q}_3 + 0 + 0 \\ 0 \\ 0 \end{bmatrix} + \begin{bmatrix} C_{12} \dot{q}_2^2 + C_{13} \dot{q}_3^2 \\ C_{21} \dot{q}_1^2 + C_{23} \dot{q}_3^2 \\ C_{31} \dot{q}_1^2 + C_{32} \dot{q}_2^2 \\ 0 \\ C_{51} \dot{q}_1^2 + C_{52} \dot{q}_2^2 \\ 0 \end{bmatrix} + \begin{bmatrix} 0 \\ g_2 \\ g_3 \\ 0 \\ g_5 \\ 0 \end{bmatrix}$$

The controller based on a formulation (59) is related to robot dynamics therefore it has problems in uncertain conditions.

### Implemented Computed Torque Controller

In first step, constructed dynamics and kinematics blocks (i.e., plant) with power supply will be put in work space. The main object is implementation of controller block. According to PD equation which is  $\ddot{q}_d + K_v \dot{e} + K_p e$ , the linearized part will be created like Figure 4. The linearized part so called PID. As it is obvious, the parameter  $e$  is the difference of actual and desired values and  $\dot{e}$  is the change of error.  $K_p$  and  $k_v$  are proportional and derivative gains and  $\ddot{q}_d$  is double derivative of the joint variable. A sample of PD controller block for one joint is like Figure 5.

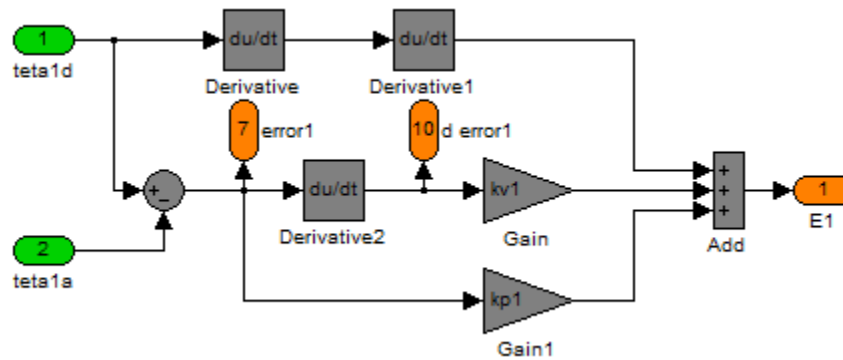
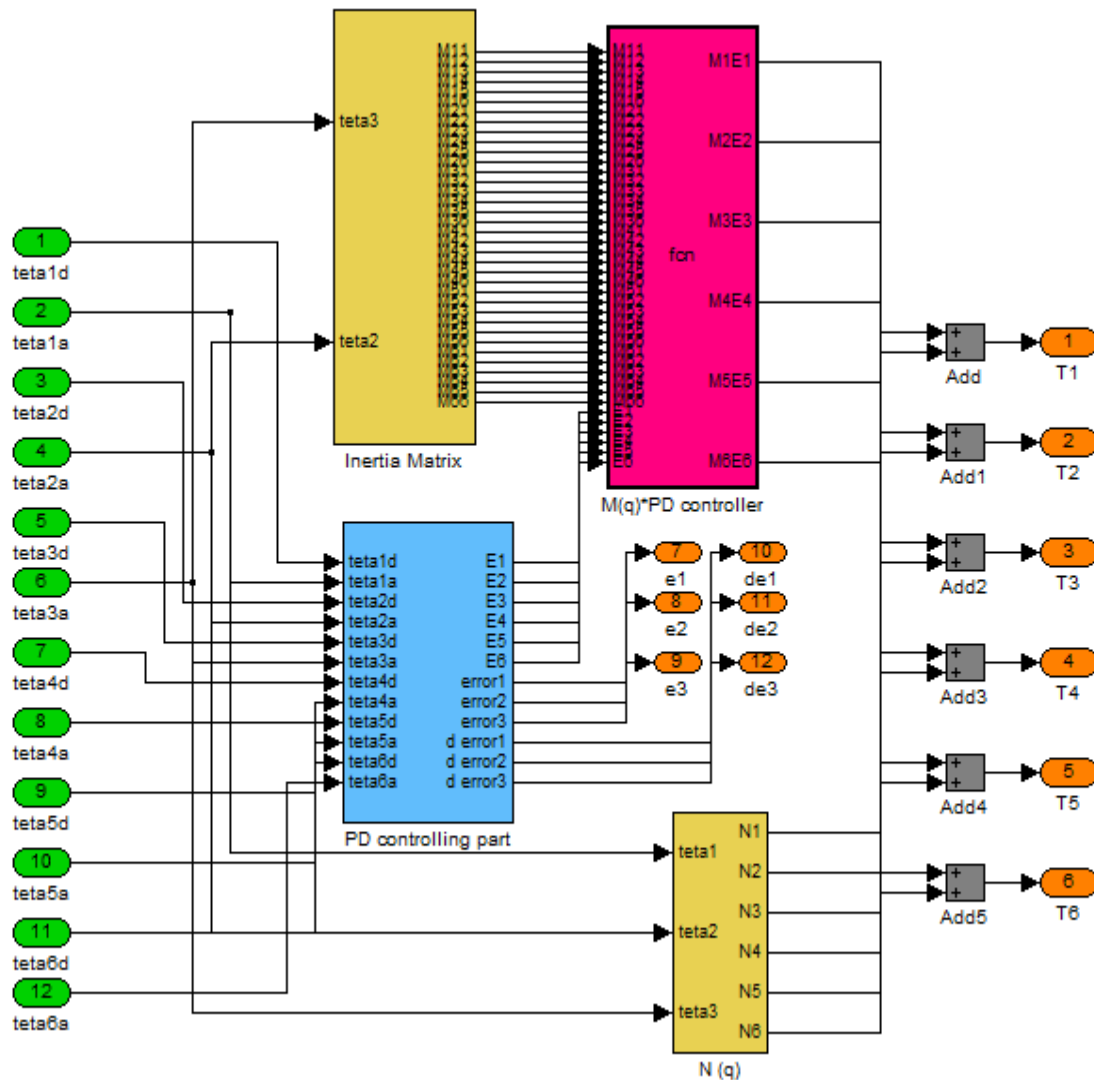


FIGURE 4: PD Controller for a joint variable

As it is seen in Figure 4 the error value and the change of error were chosen to exhibit in measurement center. In this block by changing gain values, the best control system will be applied. In the second step according to torque formulation in CTC mode, all constructed blocks just connect to each other as blew. In Figure 5 the  $N(\ddot{q}, \dot{q})$  is the dynamic parameters block (i.e., A set of Coriolis, Centrifugal and Gravity blocks).



**FIGURE 5:** The general diagram of controller

The inputs are thetas and the final outputs are torque values. The relations between other blocks are just multiplication and summation as mentioned in torque equation. In the next step transform our subsystems into a general system to form controller block and the outputs will be connected to the plant, in order to execute controlling process. Then, trigger the main inputs with power supply to check validity and performance. In Figure 6, Dynamics, Kinematics and Controller blocks are shown.

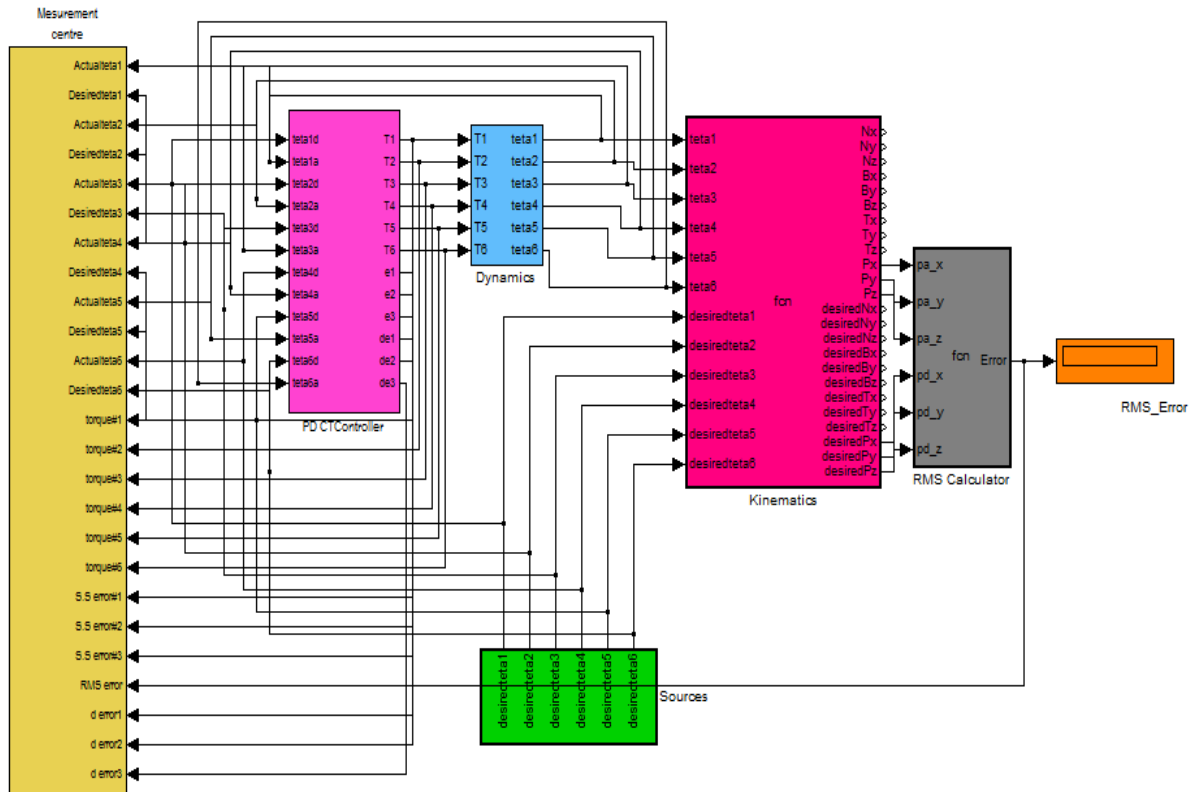


FIGURE 6 : Controller, Dynamics and Kinematics Blocks

#### 4. RESULTS

PD-Computed torque controller (PD-CTC) and PID-Computed torque controller (PID-CTC) were tested to Step and Ramp responses. In this simulation the first, second, and third joints are moved from home to final position without and with external disturbance. The simulation was implemented in MATLAB/SIMULINK environment. It is noted that, these systems are tested by band limited white noise with a predefined 40% of relative to the input signal amplitude which the sample time is equal to 0.1. This type of noise is used to external disturbance in continuous and hybrid systems.

##### Trajectory Performance

Figures 7 and 8 show tracking performance for PD-CTC and PID CTC without disturbance for two type trajectories. The optimal coefficients in Step PID CTC are;  $K_p = 70$ ,  $K_v = 15$ , and  $K_i = 75$  and in Ramp PID CTC are;  $K_p = 50$ ,  $K_v = 10$ , and  $K_i = 25$  as well as similarly in Step and Ramp PD CTC are;  $K_p = 30$  and  $K_v = 4$ ; From the simulation for first, second, and third links, it was seen that the different controller gains have the different performance. Tuning parameters of PID-CTC and PD-CTC for PUMA robot manipulator are shown in Tables 3 to 6.

	$k_{P_1}$	$k_{V_1}$	$k_{I_1}$	$k_{P_2}$	$k_{V_2}$	$k_{I_2}$	$k_{P_3}$	$k_{V_3}$	$k_{I_3}$	RMS error	SS error <sup>1</sup>	SS error <sup>2</sup>	SS error <sup>3</sup>
1	70	24	70	70	24	70	70	24	70	2.276e-5	-3.81e-5	-3.81e-5	-3.81e-5
2	50	24	70	50	24	70	50	24	70	3.34e-5	-5.6e-5	-5.6e-5	-5.6e-5
3	70	15	75	70	15	75	70	15	75	0	0	0	0
4	70	24	50	70	24	50	70	24	50	3.7e-5	-6.2e-5	-6.2e-5	-6.2e-5

**TABLE 3:** Tuning parameters of a step PID-CTC

	$k_{P_1}$	$k_{V_1}$	$k_{I_1}$	$k_{P_2}$	$k_{V_2}$	$k_{I_2}$	$k_{P_3}$	$k_{V_3}$	$k_{I_3}$	RMS error	SS error <sup>1</sup>	SS error <sup>2</sup>	SS error <sup>3</sup>
1	90	10	25	90	10	25	90	10	25	-1.2e-6	-1.6e-6	-1.6e-6	-1.6e-6
2	50	10	25	50	10	25	50	10	25	-4.5e-9	-6e-9	-6e-9	-6e-9
3	90	3	25	90	3	25	90	3	25	-8.8e-7	-1.17e-6	-1.17e-6	-1.17e-6
4	90	10	10	90	10	10	90	10	10	-2.4e-5	-3.2e-5	-3.2e-5	-3.2e-5

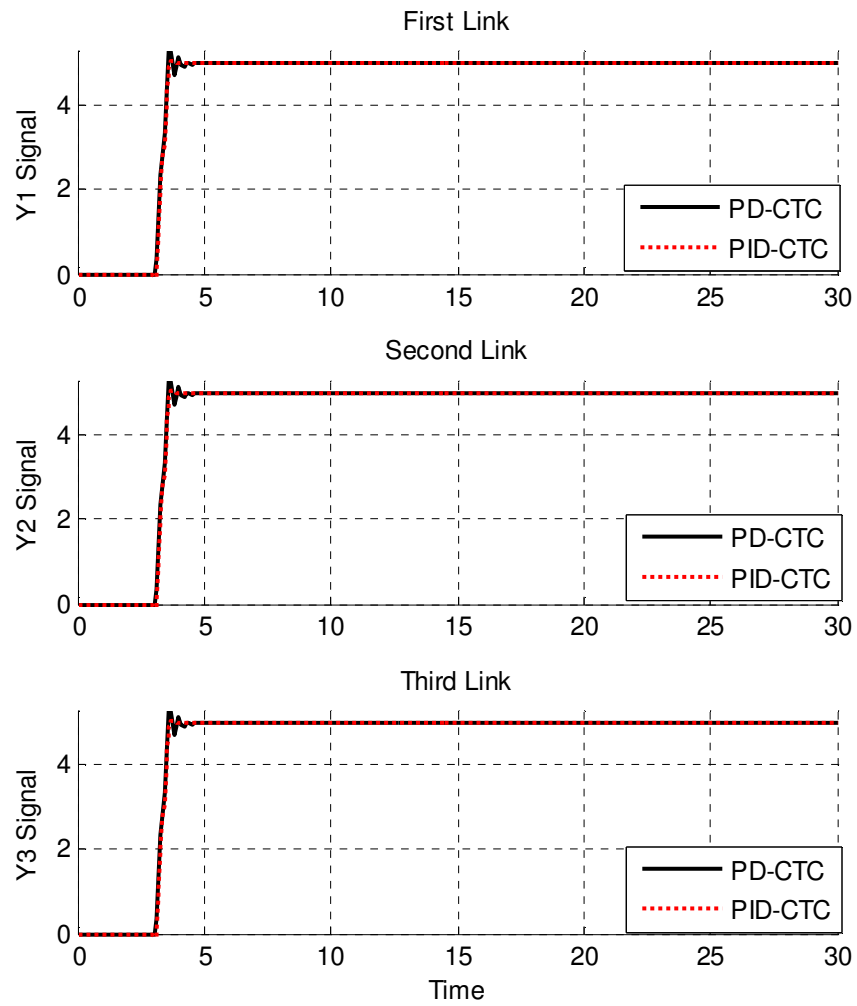
**TABLE 4:** Tuning parameters of a ramp PID-CTC

	$k_{P_1}$	$k_{V_1}$	$k_{P_2}$	$k_{V_2}$	$k_{P_3}$	$k_{V_3}$	RMS error	SS error <sup>1</sup>	SS error <sup>2</sup>	SS error <sup>3</sup>
1	8	4	8	4	8	4	2.276e-5	-3.81e-5	-3.81e-5	-3.81e-5
2	30	4	30	4	30	4	1.34e-5	-3.6e-5	-2.54e-5	-1.6e-5
3	1	4	1	4	1	4	0.0039	0.0065	0.0065	0.0065
4	8	40	8	40	8	40	0.502	5.043	5.043	5.043
5	8	0.5	8	0.5	8	0.5	0.0026	0.0043	0.0043	0.0043

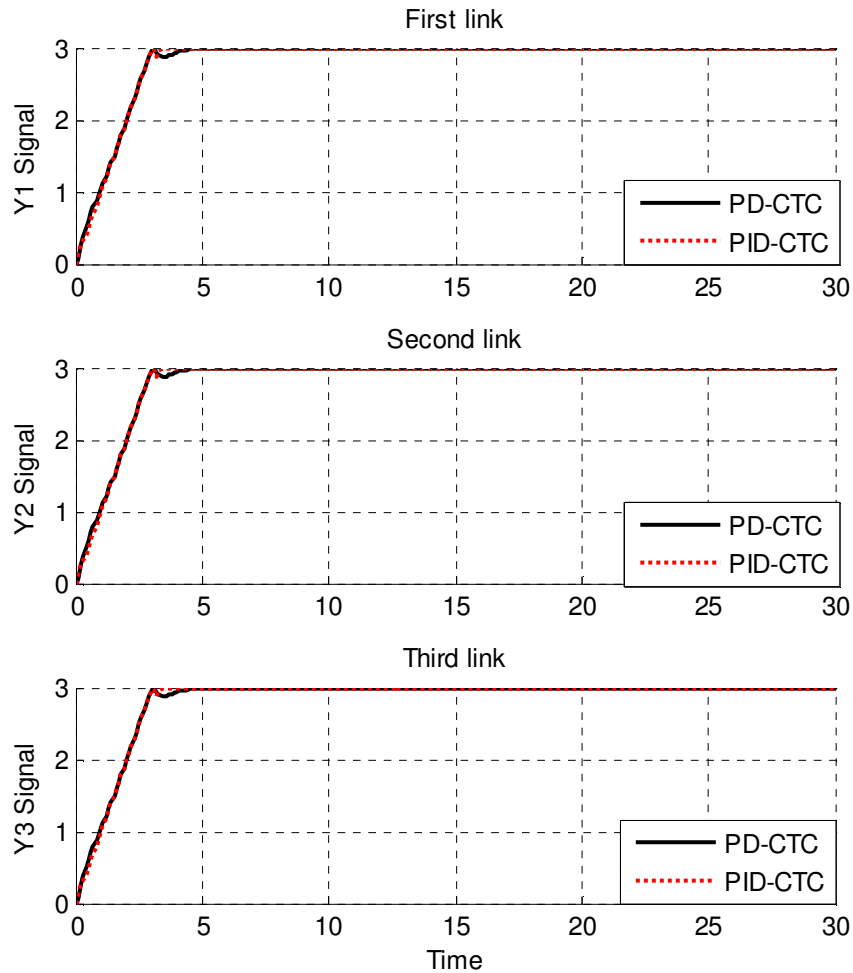
**TABLE 5:** Tuning parameters of a step PD-CTC

	$k_{P_1}$	$k_{V_1}$	$k_{P_2}$	$k_{V_2}$	$k_{P_3}$	$k_{V_3}$	RMS error	SS error <sup>1</sup>	SS error <sup>2</sup>	SS error <sup>3</sup>
1	8	4	8	4	8	4	3.2e-5	-2.81e-5	-2.6e-5	-2.1e-5
2	30	4	30	4	30	4	-6e-5	-8e-5	-8.6e-5	-8.9e-5
3	1	4	1	4	1	4	0.000305	0.00024	0.00024	0.00024
4	8	40	8	40	8	40	0.6	4.93	4.93	4.93
5	8	0.5	8	0.5	8	0.5	0.5	2.9	2.9	2.9

**TABLE 6:** Tuning parameters of a ramp PD-CTC



**FIGURE 7:** Step PID-CTC and PD-CTC for First, second and third link trajectory

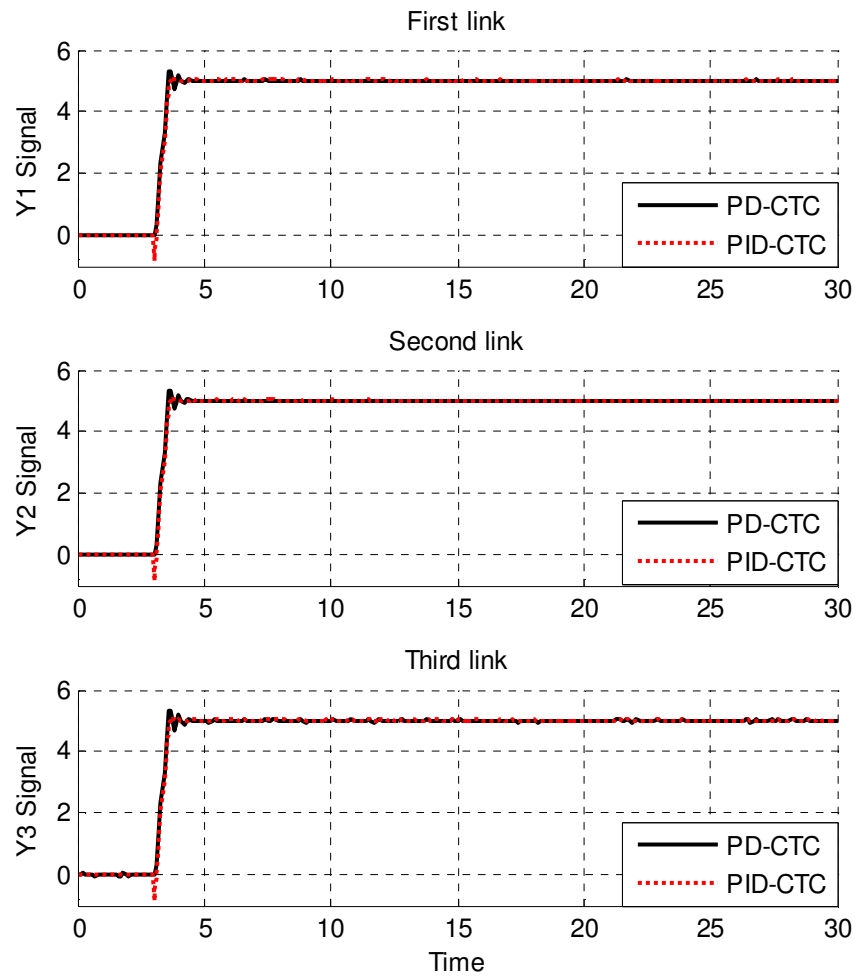


**FIGURE 8:** Ramp PID-CTC and PD-CTC for First, second and third link trajectory

By comparing step response trajectory without disturbance in PD and PID CTC, it is found that the PID's overshoot (**1.32%**) is lower than PD's (**6.44%**), although both of them have about the same rise time; PID CTC (**0.5 sec**) and PD CTC (**0.403 sec**). Besides the Steady State and RMS error in PID (**Steady State error =0 and RMS error=0**) is fairly lower than PD (**Steady State error  $\cong -3^{-5}$  and RMS error= $-1.6 \times 10^{-5}$** ).

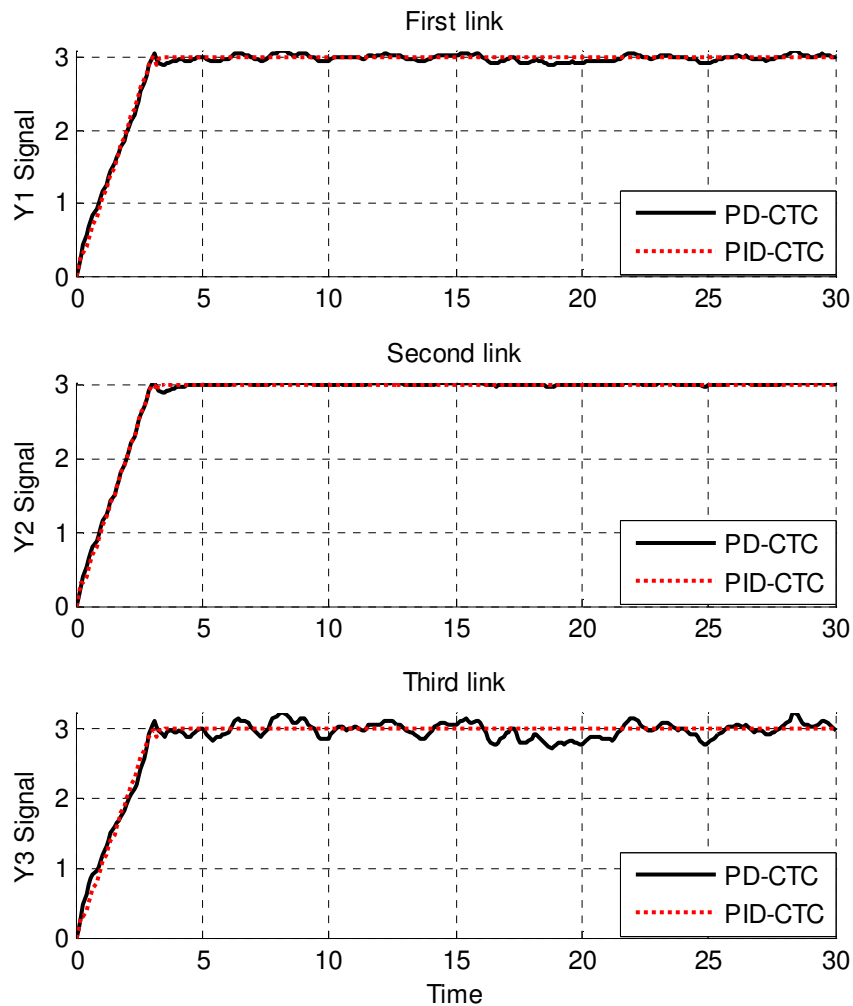
In above graphs over the same period, it is created that the PD and PID CTC are increased slightly, but PD CTC at  $t = 3 \text{ sec}$  has very low distortion, after  $t = 3 \text{ sec}$  bought of graphs remain are unchanged as mentioned to the Tables 4 and 6.

**Disturbance Rejection:** Figures 9 and 10 have shown the power disturbance elimination in PD and PID CTC. The main target in this controller is disturbance rejection as well as the other responses. A band limited white noise with predefined of 40% the power of input signal is applied to the Step and Ramp PD and PID CTC. It found fairly fluctuations in trajectory responses. As mentioned earlier, CTC works very well when all parameters are known, this challenge plays important role to select the SMC as a based robust controller in this research.



**FIGURE 9:** Step PID-CTC and PD-CTC for First, second and third link trajectory with disturbance.

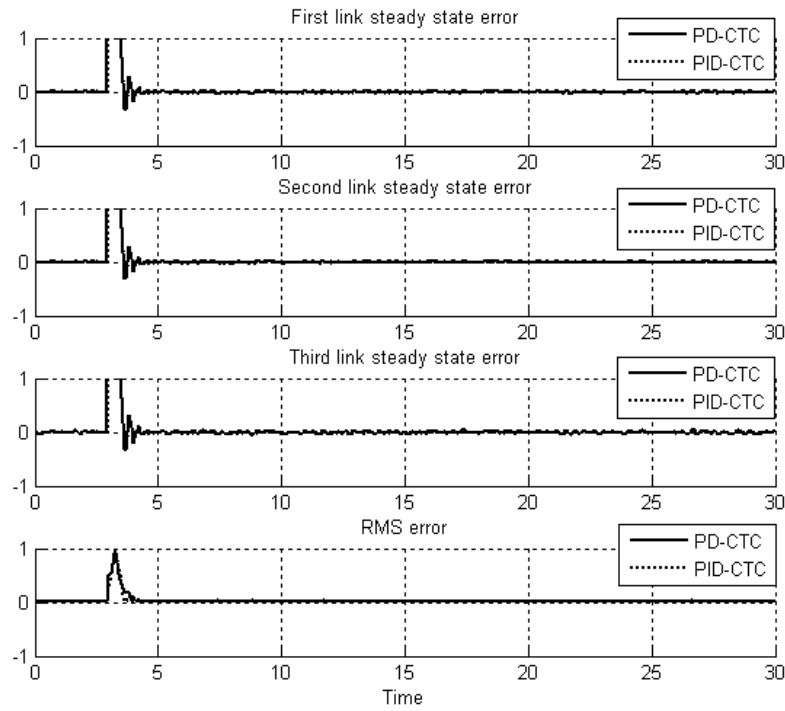




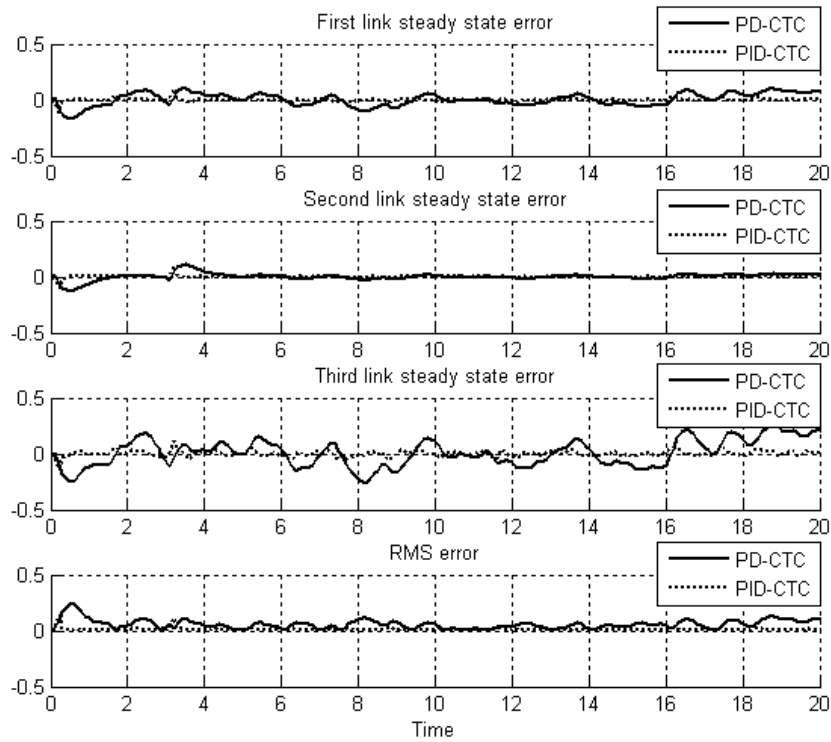
**FIGURE 10:** Ramp PID-CTC and PD-CTC for First, second and third link trajectory with disturbance.

Among above graphs (9 and 10) relating to Step and Ramp trajectories following with external disturbance, PID CTC and PD CTC have fairly fluctuations. By comparing some control parameters such as overshoot, rise time, steady state and RMS error it computed that the PID's overshoot (1.8%) is lower than PD's (8%), although both of them have about the same rise time; PID CTC (0.5 sec) and PD CTC (0.41 sec), the Steady State and RMS error in PID (Steady State error = -0.0019 and RMS error=0.0025) is fairly lower than PD (Steady State error = 0.005 and RMS error=0.0042).

**Errors in the model:** Figures 11 and 12 have shown the error disturbance in PD and PID CTC. The controllers with no external disturbances have the same error response, but PID CTC has the better steady state error when the robot manipulator has external disturbance. Furthermore the RMS error profile for PID CTC is sharply dropped compared to the PD CTC.



**FIGURE 11:** Step PID-CTC and PD-CTC for First, second and third link steady state and RMS error with disturbance.



**FIGURE 12:** Ramp PID-CTC and PD-CTC for First, second and third link steady state and RMS error with disturbance.

The errors in PID CTC and PD CTC is widely increased among of error graphs (11 and 12) relating to Step and Ramp response with external disturbance. By comparing the steady state and RMS error it observed that the PID's State and RMS error (**Steady State error = -0.0019 and RMS error=0.0025**) is lower than PD's (**Steady State error  $\cong$  0.005 and RMS error=0.0042**). When applied disturbance in these controllers it is computed that the steady state and RMS error increased rapidly approximately 130%.

## 5. CONCLUSION

In this research we introduced, basic concepts of robot manipulator (e.g., PUMA 560 robot manipulator) and control methodology. PUMA 560 robot manipulator is a 6 DOF serial robot manipulator. From the control point of view, robot manipulator divides into two main parts i.e. kinematics and dynamic parts. The dynamic parameters of this system are highly nonlinear. To control of this system nonlinear control methodology (computed torque controller) is introduced. Computed torque controller (CTC) is an influential nonlinear controller to certain and partly uncertain systems which it is based on feedback linearization and computes the required arm torques using the nonlinear feedback control law. When all dynamic and physical parameters are known computed torque controller works superbly.

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